
AIRCRAFT VERIFICATION

Date: November 8, 2011

Make & Model: Cessna 172S

N 6196P

Serial #: 172S10754

Year: 2008

Total Time Airframe: 2363.5

Total Time SMOH Engine: N/A

Total Time SMOH Propeller: N/A

Next Inspection type: 50 Hour

Due at Tach: 2413.5

Annual Inspection Due: 5/2012

E.L.T. Battery replacement Due: 5/2013

E.L.T. Check per FAR 91.207 (d) Due: May 31, 2012

Transponder check Due: 6/2012

Pitot Static System check Due: 6/2012

Altimeter check Due: 6/2012

Fire Extinguisher Visual Inspection Due: November 30, 2011

All AD's complied with through: 2011-22

Recurring AD's:

AD#	SUBJECT	NEXT DUE TACH:
AD84-26-2	Induction Air Filter	2863.5
AD93-5-6	Ignition Switch	3997.4
AD2008-14-7	Fuel Injector Lines	2463.5

All installed equipment has been verified to meet this aircraft manufacturer's currently approved equipment list, FAA approved TCDS, FAA major alteration and repair Form 337, and all items above are certified true and correct.


Kevin Newport AP521138641

Compass Deviation Log Sheet

A/C N6196P

Note

(Deviation is only valid as long as avionics type and installation setup are unchanged.)

For	N	30	60	E	120	150
Steer	360	30	60	90	120	151
For	S	210	240	W	300	330
Steer	181	211	241	270	300	330

Date 7/7/08

For	N	30	60	E	120	150
Steer						
For	S	210	240	W	300	330
Steer						

Date _____

For	N	30	60	E	120	150
Steer						
For	S	210	240	W	300	330
Steer						

Date _____

For	N	30	60	E	120	150
Steer						
For	S	210	240	W	300	330
Steer						

Date _____

1. Approving National Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number. 019078	
4. Organization Name and Address. AVSTAR FUEL SYSTEMS, INC. 1365 PARK LANE SOUTH JUPITER FLORIDA 33458				5. Work Order, Contract, or Invoice Number: 019078		
6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	RSA-5AD1 SERVO FUEL INJECTOR	2576536-2	VARIOUS	1ea	70189405	Overhauled
13. REMARKS Overhauled in accordance with FAA approved data from the manufacturer's Component Maintenance Manual (Precision Airmotive Component Maintenance Manual for Model RSA-5AD1 Servo Fuel Injectors, Form No. 15-381G, Change 1, dated 06/01/92). Pertinent details of the overhaul are on file at this repair station under the work order referenced in block 5. SEE ATTACHED FORM No.: AFS900-F6						
14. Certifies the item identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13			19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature:		16. Approval Authorization No:	20. Authorized Signature: <i>[Signature]</i>		21. Approval/Certificate No: V1RR580Y	
17. Name (Typed or Printed):		18. Date (m/d/y):	22. Name (Typed or Printed): ROBERT ATKINSON		23. Date (m/d/y): 4/10/2012	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs works in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1 it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulation by the user/installer before the aircraft may be flown.						

1 year

max 10m

4/11/2012



**BENDIX RSA-5AD1
COMPLIANCE REPORT**

DOC: AFS900-F6

REV: 3

FEBRUARY 24, 2012

**AVStar Fuel Systems, Inc.
FAA Repair Station V1RR580Y**

BENDIX FUEL SERVO RSA-5AD1

AD or SB Number	Complied	Method of Compliance	Recurring
SB RS-71	04/10/12	Apply Loctite to stem nut	Overhaul
AD 78-23-10	"	Change center body bellows seal	Overhaul
AD 79-04-05	"	Diaphragm inspection	Overhaul
SB RS-85,86,87,88	"	New fuel diaphragm	Overhaul
AD 83-22-04	"	New fuel diaphragm	Overhaul
AD 77-25-05	"	Pressure sensitivity check	Overhaul
AD 2009-02-03	"	N/A by P/N, installed 2577258 gasket	Overhaul
AD 2011-15-10 AD 2012-03-06	"	N/A by lot number of diaphragm assembly	No

Part No. 2576536-2
Model No. RSA-5AD1
Serial No. 70189405
Work Order No. 019078

Technician's Signature:

1. Approving National Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 019886	
4. Organization Name and Address: AVSTAR FUEL SYSTEMS, INC. 1365 PARK LANE SOUTH JUPITER FLORIDA 33458				5. Work Order, Contract, or Invoice Number: 019886		
6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	FLOW DIVIDER	2576564-1	VARIOUS	1ea	135	Overhauled
13. REMARKS Overhauled in accordance with FAA approved data from the manufacturer's Component Maintenance Manual (Precision Airmotive Overhaul Manual, Flow Dividers used with RS/RSA Injectors, Form No. 15-540D, dated 12/16/04). Pertinent details of the overhaul are on file at this repair station under the work order referenced in block 5. SEE ATTACHED FORM AS # AFS900-F9						
14. Certifies the item identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13			19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature:		16. Approval Authorization No:	20. Authorized Signature: <i>Mark Heermance</i>		21. Approval/Certificate No: V1RR580Y	
17. Name (Typed or Printed):		18. Date (m/d/y):	22. Name (Typed or Printed): MARK HEERMANCE		23. Date (m/d/y): 4/12/2012	
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs works in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1 it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulation by the user/installer before the aircraft may be flown.						



**BENDIX FLOW DIVIDER
COMPLIANCE REPORT**

DOC: AFS900-F9

REV: 1

November 26,2008

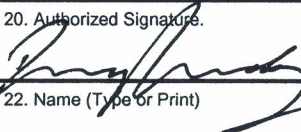
AVStar Fuel Systems, Inc.
FAA Repair Station V1RR580Y

BENDIX FLOW DIVIDER

AD or SB Number	Complied	Method of Compliance	Recurring
AD 75-09-15	4/12/2012	Replace Flow Divider Gasket	Overhaul

Part No. 2576564-1
Model No. FLOW DIVIDER
Serial No. 135
Work Order No. 019886

Technician's Signature: Mark Peckman

1. Approving National Aviation Authority / County: FAA / United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number 88363	
4. Organization Name and Address: Pacific Oil Cooler Service, Inc. 1677 Curtiss Court, La Verne, CA				F.A.A RF3R813L 91750		5. Work Order/Contact/Invoice No 88363	
6. Item	7. Description	8. Part Number	9. Eligibility *	10. Quantity	11. Serial/Batch Number	12. Status/Work:	
1	OIL COOLER	10877A	N/A	1	3883	OVERHAULED	
13. Remarks: Work Order Number (Block 5) describing the actual work performed is attached. The described work was performed in accordance with Manufacturer's Specification. (See Work Order) No applicable Airworthiness Directives work / status. Certifies that the work specified in Blocks 12/13 was carried out in accordance with EASA part 145, and with respect to that work, the component is considered ready for release to service under EASA part 145 Approval Number: EASA.145.5554							
14. Certified the items identified above were manufactured in conformity to:			19 <input checked="" type="checkbox"/> 14 CRF 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specific in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.				
15. Authorized Signature		16. Approval /Authorization No:		20. Authorized Signature: 		21. Approval/Certificate No RF3R813L	
17. Name (Type or Print)		18. Date (m/d/y):		22. Name (Type or Print) Danny Fernandez		23. Date (MMDDYYYY): May-21-2012	
User / Installer Responsibilities							
It is important that the existence of the document alone does not automatically constitute authority to install the part, component, or assembly above. When the user / installer performs the work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that that user / installer ensures that his/her airworthiness authority accepts parts, components, and or assemblies from the airworthiness authority of the country specified in Block 1. Statements in Block 14 and 19 do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

1. Approving National Aviation Authority/Country: UNITED STATES	2.	<h1 style="margin:0;">AUTHORIZED RELEASE CERTIFICATE</h1> <p style="margin:0;">FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>	3. Form Tracking Number: 35082
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4. Organization Name and Address: AERO ACCESSORIES 16152 VALERIO ST. VAN NUYS CALIF. 91406 (XB2R286)	5. Work Order/Contract/Invoice Number: 35082
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6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	MAGNETO	4371	TBV BY INSTALLER	1	09022702	REPAIRED

13. Remarks:

REPAIRED, INSPECTED, AND TESTED IN ACCORDANCE WITH SLICK SERVICE MANUAL AS PER 500HR. INSP.
 NOTE: THIS MAGNETO COMPLIES WITH SLICK S.B. 2-08
 S.B. 3-08 N.A. TEMPEST CARBON BRUSH INSTALLED.

14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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15. Authorized Signature: 	16. Approval/Authorization No.: XXXXXXXXXXXXXXXXXXXX	20. Authorized Signature: 	21. Approval/Certificate No.: XB2R286L
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17. Name (Typed or Printed): XXXXXXXXXXXXXXXXXXXXXXXXXXXX	18. Date: XXXXXXXXXXXXXXXXXXXX	22. Name (Typed or Printed): ROBERT M. RUMMM	23. Date (m/d/y): MAY/17/2012
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User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving National Aviation Authority/Country: UNITED STATES	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 35081
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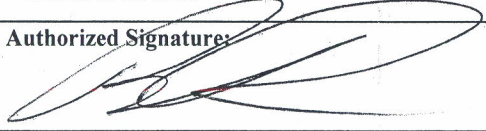
4. Organization Name and Address: AERO ACCESSORIES 16152 VALERIO ST. VAN NUYS CALIF. 91406 (XB2R286)	5. Work Order/Contract/Invoice Number: 35081
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6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	MAGNETO	4371	TBV BY INSTALLER	1	08111376	REPAIRED

13. Remarks:

REPAIRED, INSPECTED, AND TESTED IN ACCORDANCE WITH SLICK SERVICE MANUAL AS PER 500HR. INSP.
 NOTE: THIS MAGNETO COMPLIES WITH SLICK S.B. 2-08
 S.B. 3-08 N.A. TEMPEST CARBON BRUSH INSTALLED.

14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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15. Authorized Signature:	16. Approval/Authorization No.: XXXXXXXXXXXXXXXXXXXX	20. Authorized Signature: 	21. Approval/Certificate No.: XB2R286L
17. Name (Typed or Printed): XXXXXXXXXXXXXXXXXXXX	18. Date: XXXXXXXXXXXXXXXXXXXX	22. Name (Typed or Printed): ROBERT M. RUMM	23. Date (m/d/y): MAY/17/2012

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving National Aviation Authority/Country: UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: 35083	
4. Organization Name and Address: AERO ACCESSORIES 16152 VALERIO ST. VAN NUYS CALIF. 91406 (XB2R286L)					5. Work Order/Contract/Invoice Number: 35083		
6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:	
1	ALTERNATOR	9910591-11RX	TBV BY INSTALLER	1	K031260	OVERHAULED	
13. Remarks: REPAIRED, INSPECTED & TESTED IN ACCORDANCE WITH KELLY AEROSPACE SERVICE MANUAL AS PER OVERHAUL.							
14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.				19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature:		16. Approval/Authorization No.: XXXXXXXXXXXXXXXXXXXX		20. Authorized Signature: 		21. Approval/Certificate No.: XB2R286L	
17. Name (Typed or Printed): XXXXXXXXXXXXXXXXXXXX		18. Date: XXXXXXXXXXXXXXXXXXXX		22. Name (Typed or Printed): ROBERT M. RUMMM		23. Date (m/d/y): MAY/17/2012	
User/Installer Responsibilities							
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.							
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.							
Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

1 Approving National Aviation Authority / Country: FAA / UNITED STATES	2 AUTHORIZED RELEASE CERTIFICATE FAA FROM 8130-3, AIRWORTHINESS APPROVAL TAG	3 Form Tracking Number: AA I7281-0000
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4. Organization / Name and Address Aero Accessories, Inc. 1240 Springwood Avenue, Gibsonville, NC	5. Work Order / Contract / Invoice Number: I7281-0000
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6. Item	7. Description	8. Part Number	9. Eligibility	10. Quantity	11. Serial / Batch Number	12. Status / Work
1	Dry Air Pump	215CC	N/A	1	I728104	OVERHAUL

13 Remarks OVERHAUL IAW ONE OF THE FOLLOWING FAA APPROVED PROCESSES: 203-6-1009, 203-6-1010, 203-6-1011, 203-6-1012 CERTIFIES THAT THE WORK SPECIFIED IN BLOCKS 12/13 WAS CARRIED OUT IAW FAR 145 AND IN RESPECT TO THAT WORK THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE FOR WORK PERFORMED UNDER EASA ACCEPTANCE CERTIFICATE NUMBER EASA 145.5682

	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return To Service <input checked="" type="checkbox"/> Other Regulation Specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in block 12 and described in block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.	
	20. Authorized Signature <i>Mitchell Murphy</i>	21. Approval/Certificate No: YV4R688M
	22. Name (Typed or Printed) Mitch Murphy	23. Date (M / D / Y) 09/15/2011

User / Installer Responsibilities
It is important to understand that: the existence of this document alone does not automatically constitute authority to install the part / component / assembly.
Where the user / installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in Block 1. It is essential that the user/installer ensures that his/her Airworthiness Authority accepts parts / components / assemblies from the Airworthiness Authority of the country specified in block 1.
Statements in block 14 and 19 do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user / installer before the aircraft may be flown.

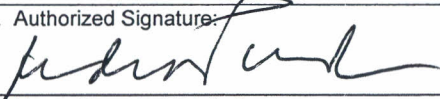
1. Approving National Aviation Authority/Country: FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: K28267
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4. Organization Name and Address: SANTA MONICA PROPELLER SERVICE, INC 3135 DONALD DOUGLAS LOOP SOUTH SANTA MONICA, CA 90405	FAA ARS ZY2R362L	5. Work Order, Contract, or Invoice Number: 82383
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6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	MCCAULEY PROPELLER	1A170E/JHA7660	N/A	1	ABK23518A	OVERHAULED

13. REMARKS
 FIXED PITCH PROPELLER ASSEMBLY WAS COMPLETELY OVERHAULED. ANGLES AND ALIGNMENT CHECKED. PROPELLER PENETRANT INSPECTED, COMPLIED WITH SB 240E, REFINISHED, ALODINED, PAINTED AND BALANCED I/AW SM 730720 REV.6. NEXT OVERHAUL DUE AT 2000 FLIGHT HOURS OF OPERATION OR 72 CALENDAR MONTHS, WHICHEVER OCCURS FIRST. NEXT ACTION DUE AT 1000 HOURS.
 -----END-----

14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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15. Authorized Signature:	16. Approval Authorization No:	20. Authorized Signature: 	21. Approval/Certificate No: ZY2R362L
17. Name (Typed or Printed):	18. Date (m/d/y):	22. Name (Typed or Printed): PEDRO PAREDES	23. Date (M D Y): MAY 29, 2012

User/Installer Responsibilities

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1 it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

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WORK ORDER



GERDES AVIATION SERVICES

FAA APPROVED REPAIR STATION G3RR399J

CUSTOMER TOM'S A/C - JEFF WORK ORDER/INV# 8879A
 AIRCRAFT N# 6196P TACH/HOBBS 2446.0 ACTT 2446.0
 MAKE CESSNA MODEL 172S SN# 172S10754 (2008)
 OWNER NAME TOM'S AIRCRAFT MAINT, INC
 ADDRESS 2641 E. SPRING STREET
 CITY LONG BEACH STATE CA ZIP 90806-2218
 PHONE 1 (8) 562-426-5913 PHONE 2 (8) 562-427-8236
 EMAIL TOM@TOMSAIRCRAFT.COM

WORK REQUESTED

- Altimeter / Static / Transponder Transponder Mode S Test
 Leak / Correlation Test Repair Retest Warranty
 Part 135 per attached P.O.

#1 #2 PRELIMINARY INSPECTION

- Pitot / Static System Leakage / Blockage / Inoperative
 Encoder / Transponder - Inoperative / Erroneous Altitude
 Altimeter Out Of Calibration / Friction
 Other GARMIN G1000 GPN# 190-00384-12
 SYS ID: 250FC0480 CONFIG ID: (7ER6466E)
 Notes SYS SWVER: (0563.26)
 GARMIN GTX-33
 No Discrepancies Noted

#1 #2 TRANSPONDER TESTS (14 CFR 43 APP. F) (REF. 91.413)

- Reply freq 1090 ± 3.0 MHz
 Suppression P2=P1 (Replies 1% or less)
 P2 9 DB less than P1 (Replies at least 90%)
 Receiver Sensitivity (MTL) -73 ± 4 (DBM)
 Diff. between mode A & mode C not more than 1 (DBM)
 RF peak output - at least (125 W) not more than (500 W)
 Code switches display correct codes as selected
 Ident. & light operation
 Mode S Tests 14 CFR 43 APP. F (e) - (j)

ALTIMETER/STATIC SYSTEMS TESTS (14 CFR 43 APP. E)

(TABLE 1)			CALIBRATION (FEET)			(REF. 91.411)		
READING	ERROR #1 #2	TOL ±	READING	ERROR #1 #2	TOL ±	READING	ERROR #1 #2	TOL ±
-1,000	-12 -10	20	3,000	-23 +5	30	16,000	-30 +5	110
-500	- -	20	4,000	-30 -5	35	18,000	-42 +30	120
0	-17 0	20	6,000	-26 -15	40	20,000	-47 +60	130
500	-20 0	20	8,000	-27 -20	60	22,000		140
1,000	-19 +10	20	10,000	-27 -30	80	25,000		155
1,500	-22 0	25	12,000	-27 -15	90	30,000		180
2,000	-23 +10	30	14,000	-27 0	100	35,000		205

- #1 #2
 Static System Leakage Less Than 100 ft/min @ 1,000 ft AGL
 Static Port Heater N/A

TEST RESULTS

Altimeter / Static Tests Pass Fail
 #1
 #2
 Certified to 20,000 ft

- (TABLE 2)
 Hysteresis 50%
 Hysteresis 40%
 After Effect

Transponder Tests Pass Fail
 #1
 #2

- (TABLE 3)
 Friction Test
 Case Leak @ 18,000 ft
 (TABLE 4) Baro Scale Error
 Altimeter / Encoder Correlation - Not to exceed 125 ft

Mode S Tests Pass Fail
 #1
 #2

CORRECTIVE ACTION TAKEN NONE

DESCRIPTION	QTY	RATE	AMOUNT

PARTS USED NONE

PART NUMBER	DESCRIPTION	QTY	RATE	AMOUNT

This aircraft has been tested/repared in accordance with current FAA regulations and is approved for return to service. TOTAL _____
 N/A Work not completed on this date.
 Part 91 Part 135 Data Source / Rev # / Date _____

Authorized Signature Brian Inoué Date 06-19-2012 Form WO-2009-R1
 Office: 714-525-7545 • PO Box 2872 • Fullerton CA 92837 • FAA Approved Repair Station G3RR399J © 2009 All rights reserved by Gerdes Aviation Services

AVIONICS INSTALLATIONS, INC.
100 Cessna Blvd, Ste 1A
Port Orange, FL 32128
386-492-7760

CRS G43R643N

WORK ORDER NUMBER 10426

CUSTOMER NAME: HALCYON FLIGHT CLUB

DATE: 6-14-10

ADDRESS: PO BOX 771 WINTER PARK FL 32790

DAY PHONE: 478-8715 EPIC

HOME PHONE: _____

A/C NUMBER: N6196P

A/C MAKE: Cessna

A/C MODEL: 172S

A/C S/N: 172S10254

APPLIANCE: Transponder

ACT

ADC

MANUFACTURER'S NAME: GARMIN

United

GARMIN

MODEL #: GTX33

5934PD-3

GDC74

SERIAL #: 81107027

467146

20601760

IDENTIFICATION OF ITEM WORKED ON OR WORK REQUESTED

IFR Cert

PRELIMINARY INSPECTION: DATE 6-14-10

OK

PERFORMED BY: Gray Rickett

HIDDEN DAMAGE INSPECTION: DATE 6-14-10

OK

PERFORMED BY: Gray Rickett

DETAILED DESCRIPTION OF WORK PERFORMED

performed IFR Cert

The ATC Transponder test required by FAR 91.413 was

The Altimeter/Altitude Reporting/Static System Test required

Performed and complies with FAR 43 Appendix F

by FAR 91.411 was performed and complies with FAR 43

Appendix E

Performed By: Gray Rickett

Date

6-14-10

Inspected By: Gray Rickett

Date

6-14-10

MAINTENANCE RELEASE

THE ITEMS AS IDENTIFIED ABOVE WERE INSPECTED/CERTIFIED IN ACCORDANCE WITH CURRENT MAINTENANCE RULES OF THE FEDERAL AVIATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE.

DATE: 6-14-10

SIGNED: Gray Rickett

Pertinent details of the repair are on file at this Repair Station.

At no time will an INSPECTION STAMP/SYMBOL be used in the signature block of the maintenance release.

ALTIMETER CERTIFICATION TEST FORM

AVIONICS INSTALLATIONS, INC.
100 CESSNA BLVD, STE 1A
PORT ORANGE, FL 32128
(386) 492-7760
CRS G43R643N

ALTIMETER CERTIFICATION TEST FAA Certified Repair Station

Altimeter #1	P/N	<u>5934PD-3</u>	S/N	<u>467146</u>	W.O. No. <u>10426</u>
Altimeter #2	P/N		S/N		
Remote Encoder	P/N	<u>GDC 74A</u>	S/N	<u>20601760</u>	
Air Data Computer	P/N		S/N		

Altitude	Actual Reading (1 minute min)	± Tol	Friction	± Tol	Encoder	± Tol
-1,000	-1000	20	-	-	-	-
0	-5	20	-	-	-	-
500	500	20	-	-	-	-
1,000	1000	20	15	70	-	-
1,500	1505	25	-	-	-	-
2,000	2005	30	15	70	2000	125
2,200	-	-	-	-	2700	125
2,700	-	-	-	-	2700	125
3,000	3000	30	20	70	3000	125
4,000	3985	35	-	-	-	-
5,000	-	-	25	70	5000	125
6,000	5995	40	20	70	-	-
8,000	7995	60	-	-	8000	125
10,000	9995	80	25	80	-	-
12,000	11975	90	-	-	-	-
14,000	13970	100	-	-	-	-
15,000	-	-	30	90	-	-
16,000	15970	110	30	90	16000	125
18,000	17960	120	-	-	-	-
20,000	19960	130	35	100	20000	125
22,000		140	-	-	-	-
25,000	X	155	X	120	X	-
30,000	X	180	X	140	X	-
35,000	X	205	X	160	X	125
40,000	X	230	X	180	X	-
45,000	X	255	X	-	X	-
50,000	X	280	X	250	X	125

Pressure (in of HG)	Alt. Differential (feet)	Alt. Reading Tol ± 25 ft
28.10	-1,727	-1720
28.50	-1,340	-1335
29.00	-863	-850
29.50	-392	-385
29.92	0	-5
30.50	±531	525
30.90	±893	880
30.99	±974	970

Case Leak at 18,000 ft
(±100 ft max.) 15

Hysteresis Test No. 1
(50%) ±75 ft 25

Hysteresis Test No. 2
(40%) ±75 ft 20

After Effect Test (±30 ft) 10

TRANSPONDER CERTIFICATION FORM

AVIONICS INSTALLATIONS, INC.
100 CESSNA BLVD, STE 1A
PORT ORANGE, FL 32128
(386) 492-7760

Frequency (1090±3MHZ)	<u>1090</u>	W.O.#	<u>10426</u>
Mode-S Check	<u>OK</u>	Make:	<u>GARMIN</u>
Ident (Seconds)	<u>OK 25</u> (20-30)	Model	<u>GT33</u>
Mode A Codes	<u>OK</u>	S/N	<u>89107027</u>
Mode C Codes	<u>OK</u>	N#	<u>6196P</u>
(If applicable attach Altimeter Certification Test)			
Sensitivity (90% reply)	<u>73</u>	73 ± 4	.dbm Mode A
(90% reply)	<u>72</u>		.dbm Mode C
(Within one db of previous reading.)			
Decoding			
Mode A	<u>OK</u>		
Mode C	<u>OK</u>		
Framing	<u>OK</u>		
SLS	<u>OK</u>		
Power	<u>450</u>		watts (125-500)
Correlation in A/C	<u>yes</u>		

Frequency (1090±3MHZ)	_____	W.O.#	_____
Mode-S Check	_____	Make:	_____
Ident (Seconds)	_____ (20-30)	Model	_____
Mode A Codes	_____	S/N	_____
Mode C Codes	_____	N#	_____
(If applicable attach Altimeter Certification Test)			
Sensitivity (90% reply)	_____	73 ± 4	.dbm Mode A
(90% reply)	_____		.dbm Mode C
(Within one db of previous reading.)			
Decoding			
Mode A	_____		
Mode C	_____		
Framing	_____		
SLS	_____		
Power	_____		watts (125-500)
Correlation in A/C	_____		



Applicable Airworthiness Directives

ID No: 26167

Page No: 1

Model: 172S, S/N: 172S10754, Tail No: N6196P

Date: 08/07/08

AD NUMBER	TYPE*	SUBJECT	Supersedes AD Number	N/A**
		ENGINE(s): IO-360-L2A		
		PROPELLER(s): 1A170E/JHA		
		MAGNETO(s): Slick		
84-26-02	R	INDUCTION AIR FILTER		
93-05-06	R	IGNITION SWITCH		
2001-06-17	NM	OVER-RICH MIXTURE CONDITION		
2004-10-14	R	LYCOMING ENGINE	91-14-22	
2007-04-19R1	N	SUPERIOR CYLINDERS		
2008-02-18	N	BALLISTIC RECOVERY SYSTEMS		
2008-08-14	R	PRECISION AIRMOTIVE FUEL INJECTORS	2008-06-51	
2008-10-02	NM	ALTERNATE STATIC SOURCE VALVE		
2008-14-07	R	LYCOMING ENGINE	2002-26-01	
		AD 11-26-04		

FAX NO: (631) 765-9359

TEL: (631) 765-9375

TEL: (800) 235-6444

*Type of AD (adlog subscribers—this designation appears in the upper right corner of each adNote page):
N—Non-Repetitive. R—Repetitive. NM—Non-Repetitive, but has more than one compliance requirement.
NR—Can be either Repetitive or Non-Repetitive, depending on the method of compliance.

**Not applicable.

3/11/08

652 Oliver Street Williamsport, PA 17701 U.S.A.

Engine Model No: **IO-360-L2A**Engine SN: **L-34166-51E**

Page 2 of 2

AD NO. / SB	RV	Description / Method of Compliance	Next Due	Once Rec	Code	Sign
2006-20-09 569	A	CRANKSHAFT REPLACEMENT REPLACE CRANKSHAFT	N/A	X	2	
64-16-05 298	Inactive	AC FUEL PUMP OIL SEAL NEW CONF. FUEL PUMP INSTALLED	N/A	X	2	
66-06-03 302	A	REPLACEMENT OF CONNECTING ROD ASSY. NEW ROD CONFIGURATION	N/A	X	2	
66-20-04 307		OIL FILTER ADAPTER GASKET NEW CONF. GASKET INSTALLED	N/A	X	2	
67-22-06 305	4 B	REPLACEMENT OF BENDIX FUEL DIAPHRAGM ASSY NEW OR NEWLY O/H INSTALLED	N/A	X	2	
78-23-10 428		BENDIX FUEL INJ. BULLETIN RS-42 NEW P/N INSTALLED	N/A	X	2	
79-04-05 433	A	BENDIX FUEL INJ. BULLETIN RS-57 NEW P/N INSTALLED	N/A	X	2	
87-10-06 477	R1 A	INSPECTION AND REWORK OF LW-18790 ROCKER ARM ASSY NEW CONFIGURATION P/N INST.	N/A	X	2	
90-04-06 488	R1 A	PROPELLER GOVERNOR LINE SUPPORT NEW LINE W/STEEL CON. NUTS INS	N/A	X	2	
92-12-05 501	B	LW-14077 PISTON PIN NEW PARTS INSTALLED	N/A	X	2	
93-14-15 N/A		INST OF PLCKRD DENOTING OPER. NOT TO EXCEED 33.#HG FIELD COMPLIANCE AFTER STC	N/A	X	2	
95-07-01 N/A		CONNECTING ROD BOLT FAILURE LYCOMING PART INSTALLED	N/A	X	2	
95-26-02 398		ENGINES OPERATED WITH LOW OCTANE FUEL ENGINE IS NEW OR HAS BEEN REBUILT AT MANUFACTURER	N/A	X	2	
98-17-11 N/A		CRANKSHAFT REPAIRED BY NELSON BALANCING SERVICE MAG/PARTICLE INSP OF C/SHAFT	N/A	X	2	

Codes:

- 1 AD Applicable to Engine
- 2 AD Not Applicable to Engine
- 3 Field Compliance where applicable



AIRFRAME

Company
 Manufacturer CESSNA AIRCRAFT CORP.
 Model 172S
 Serial # 172S10754 Tach Time 1864.5
 A/C Cert. Date 6/25/08 Total Time 1864.5

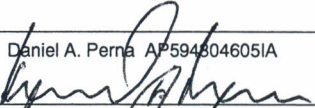
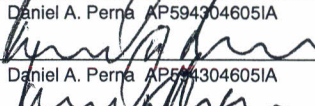
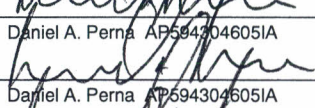
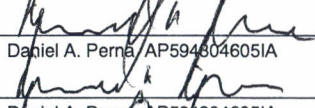
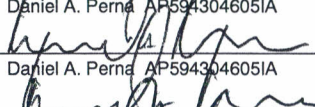
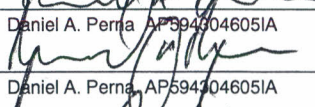
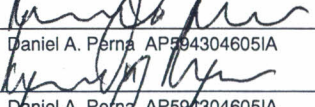
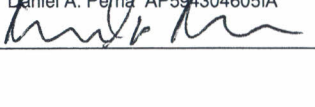



Tail Number N6196P

AD# Effective	Subject	Date & Hours @ Compliance	Method Of Compliance	O	R	Next Comp @ Hrs/Date	Authorized Signature & Number
*01-06-17 04/20/01	AFM - IMPROPER FUEL FLOW SETTINGS/	5/16/11	CW by functional check of idle / mixture		X	At idle / mixture adjustment	Daniel A. Perna AP5943046051A
03-24-13 01/20/04	SUPERSEDED BY AD 2004-15-18/	5/16/11	Superseded by AD 2004-15-18	X		N/A	Daniel A. Perna AP5943046051A
04-15-18 09/12/04	AFM, KAP140 AUTOPILOT COMPUTER SOFTWARE/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
05-05-53 03/04/05	REVISED BY AD 2005-05-53 R1/	5/16/11	Revised by AD 2005-05-53 R1	X		N/A	Daniel A. Perna AP5943046051A
05-05-53 R1 03/21/05	FLIGHT CONTROL SYSTEM RIGGING/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
05-13-10 08/09/05	MAIN ELECTRICAL POWER JUNCTION BOX CIRCUIT BREAKERS/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
06-17-04 09/01/06	SUPERSEDED BY AD 2007-08-03/	5/16/11	Superseded by AD 2007-08-03	X		N/A	Daniel A. Perna AP5943046051A
07-05-10 04/11/07	CREW SEAT BACK CYLINDER LOCK ASSEMBLY/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
07-08-03 05/02/07	ENGINE COMPARTMENT FLEXIBLE FUEL HOSE END FITTINGS/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
08-02-18 02/28/08	PARACHUTE SYSTEM PICK-UP COLLAR SUPPORT AND SCREWS/	5/16/11	N/A STC not installed	X		N/A	Daniel A. Perna AP5943046051A
08-03-02 03/06/08	FUEL RETURN LINE TO STEERING TUBE CHAFING/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
08-05-09 04/08/08	SEAT BASE/BACK ATTACH BRACKET/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
08-10-02 05/12/08	ALTERNATE STATIC AIR SOURCE SELECTOR VALVE/	5/16/11	N/A by S/N	X		N/A	Daniel A. Perna AP5943046051A
08-26-10 01/05/09	PLACARD, ALTERNATE STATIC AIR SOURCE SELECTOR VALVE ID PLACARD/	6/10/10	PCW by inspection of alternate static valve	X		N/A	Daniel A. Perna AP5943046051A
*11-06-02 05/26/11	AFM, ELECTRICAL POWER LOSS TO THE FADEC (STC SA01303W1)/	5/16/11	N/A STC not installed		X	N/A	Daniel A. Perna AP5943046051A

[Handwritten Signature]
 Date 5/10/11

ENGINE – Front

Company
 Manufacturer TEXTRON LYCOMING
 Model IO-360-L2A
 Location Front
 Serial # L-34166-51E
 TSMOH
 Total Time 1864.5

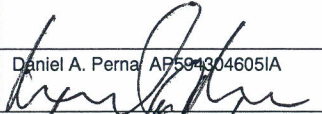
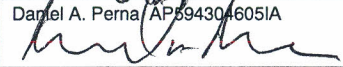
AD# Effective	Subject	Date & Hours @ Compliance	Method Of Compliance	O	R	Next Comp @ Hrs/Date	Authorized Signature & Number
*04-10-14 C1 06/25/04	CRANKSHAFT GEAR AND GEAR RETAINING BOLT/	5/16/11	N/A no propeller strike or sudden stoppage		X	At propeller strike or sudden stoppage	Daniel A. Perna AP594304605IA 
05-19-11 10/21/05	300 HORSEPOWER OR LOWER CRANKSHAFTS/	5/16/11	N/A by model	X		N/A	Daniel A. Perna AP594304605IA 
05-26-10 01/31/06	SUPERSEDED BY AD 2006-12-07/	5/16/11	Superseded by AD 2006-12-07	X		N/A	Daniel A. Perna AP594304605IA 
06-06-16 04/27/06	CRANKSHAFT REPLACEMENT/	5/16/11	N/A by model	X		N/A	Daniel A. Perna AP594304605IA 
06-10-21 06/22/06	ENGINE COMPONENTS, INC. CONNECTING RODS/	5/16/11	N/A no ECI connecting rods installed	X		N/A	Daniel A. Perna AP594304605IA 
06-12-07 07/11/06	ENGINE COMPONENTS, INC. P/N AEL65102 "CLASSIC CAST" CYLINDERS/	5/16/11	N/A no ECI cylinders installed	X		N/A	Daniel A. Perna AP594304605IA 
06-20-09 11/03/06	CRANKSHAFT RETIREMENT/	5/16/11	N/A by model	X		N/A	Daniel A. Perna AP594304605IA 
07-04-19 03/12/07	REVISED BY AD 2007-04-19 R1/	5/16/11	Revised by AD 2007-04-19 R1	X		N/A	Daniel A. Perna AP594304605IA 
07-04-19 R1 05/07/07	SUPERIOR AIR PARTS, INC. CAST CYLINDER ASSEMBLIES/	5/16/11	N/A no Superior Air Parts Inc. cylinders installed	X		N/A	Daniel A. Perna AP594304605IA 
*08-14-07 08/14/08	EXTERNAL FUEL INJECTION LINES/	5/16/11	CW by visual inspection of fuel injection lines		X	Tach: 1964.5	Daniel A. Perna AP594304605IA 
*08-19-05 10/20/08	SUPERSEDED BY AD 2009-26-12/	5/16/11	Superseded by AD 2009-26-12		X	N/A	Daniel A. Perna AP594304605IA 
*09-26-12 02/04/10	ENGINE COMPONENTS, INC. P/N AEL65102 "TITAN" CYLINDERS/	5/16/11	N/A no ECI cylinders installed		X	N/A	Daniel A. Perna AP594304605IA

* indicates possible recurring inspections © 1995 – 2007 - Tdata, Inc. (Tdata.com)

Prepared by  Date 5/16/11

PROPELLER - Front

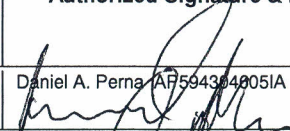
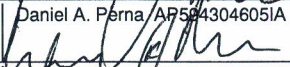
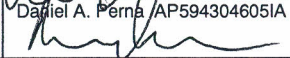
Company
Manufacturer MCCAULEY INDUSTRIAL CORP.
Model 1A170E/JHA7660
Location Front **Blade**
Tail # N6196P **TSMOH**
Serial # ABK23518A **Total Time** 1864.5 **Tail Number** N6196P

AD# Effective	Subject	Date & Hours @ Compliance	Method Of Compliance	O	R	Next Comp @ Hrs/Date	Authorized Signature & Number
03-13-17 07/18/03	MAINTENANCE REPAIR BY T AND W PROPELLERS, INC./	5/16/11	N/A no work performed by T&W Propellers	X		N/A	Daniel A. Perna AP594304605IA 
05-14-11 08/17/05	MAINTENANCE AND REPAIR BY SOUTHERN CALIFORNIA PROPELLER SERVICE/	5/16/11	N/A no work performed by Southern California Propeller Service	X		N/A	Daniel A. Perna AP594304605IA 


APPLIANCES

Company

Tail # N6196P

AD# Effective	Subject	Date & Hours @ Compliance	Method Of Compliance	O	R	Next Comp @ Hrs/Date	Authorized Signature & Number
*84-26-02 01/29/85	PAPER INDUCTION AIR FILTERS/	5/16/11	CW by replacement of filter		X	Tach: 2364.5	Daniel A. Perna / AP594304605IA 
*93-05-06 04/29/93	IGNITION SWITCHES/	5/16/11	Installed new at factory		X	Tach: 2000.0	Daniel A. Perna / AP594304605IA 
*09-02-03 02/09/09	FUEL INJECTOR SERVO PLUG AND GASKET/	6/10/10	PCW by inspection of hex plug for letter "G" stamped on face of plug		X	N/A	Daniel A. Perna / AP594304605IA 

* indicates possible recurring inspections © 1995 – 2007 - Tdata, Inc. (Tdata.com)

Prepared by  Date 5/16/11

3/11/08

652 Oliver Street

Williamsport, PA 17701 U.S.A.

Engine Model No: **IO-360-L2A**Engine SN: **L-34166-51E**

Page 1 of 2

AD NO. / SB	RV	Description / Method of Compliance	Next Due	Once	Rec	Code	Sign
2002-26-01 342	E	FUEL LINE AND SUPPORT CLAMP INSPECTION & INSTALL. NEW LINES INST. WITH NEW CLAMP	100 HOURS		X	1	3
2004-10-14 475	C	CRANKSHAFT GEAR MODIFICATION AND ASSY PROCEDURES NEW P/N INSTALLED	N/A	X		1	3
2005-19-11 566		Crankshaft replacement Replace Crankshaft	N/A	X		1	3
71-05-02 327	C	INSPECTION OF CENTER MAIN BEARING NEW INSTALLED	N/A	X		1	3
71-11-02 328		REPLACEMENT OF EXH & INT HYD TAPPET PLUNGERS NEW P/N HYD TAPPETS INSTALLED	N/A	X		1	3
73-23-01 367	F	INSP FOR CRACKS IN PISTON PINS INST NEW PARTS	N/A	X		1	3
75-09-15 382		BENDIX FUEL INJ RS-43 INSP OF MOD OF FLOW DIVIDERS NEW P/N GASKET INSTALLED	N/A	X		1	3
93-11-11 N/A		AC, TEXTRON LYC FUEL PUMP O/H BY AERO ACCESSORIES NEW AC F/P INST. WHEN APPLIC.	N/A	X		1	3
94-01-03 N/A	R2	DEFECTIVE IGN COILS AND ROTATING MAGNETS NEW CONFIGURATION PART INST.	N/A	X		1	3
94-06-09 517		INCOMPLETE INTERNAL GROUNDING ON MAG CAPACITORS NEW CONF. CAPACITORS INSTALLED	N/A	X		1	3
96-09-10 524		REPLACEMENT OF OIL PUMP IMPELLERS STEEL IMPELLERS INSTALLED	N/A	X		1	3
96-23-03 525	A	HIGH PRESSURE FUEL PUMP INSTALLED LW-15473 NEWLY MANUFACTURED PUMPS INST.	N/A	X		1	3
97-15-11 527	C	RECALL OF PISTON PIN P/N LW-14077 NEW CONFIGURATION P/N INST.	N/A	X		1	3
98-02-08 530	A	INSPECTION OF CRANKSHAFT ID FOR CORROSION MAG INSP./APPLY URETHABOND 104	N/A	X		1	3
2002-12-07 543	B	OIL FILTER CONVERTER PLATE GASKET REPLACEMENT NEW P/N CONVERTER PLATE INSTALLED	N/A	X		2	
2003-14-03 529	B	ROTARY FUEL PUMP TORQUE. NEW PUMP WITH "M" SUFFIX INSTALLED	N/A		X	2	
2006-06-16 566 Sup 1		Replace crankshaft Crankshaft replaced	N/A	X		2	

Codes:

- 1 AD Applicable to Engine
- 2 AD Not Applicable to Engine
- 3 Field Compliance where applicable



SB	RV	Method of Compliance			
2006-20-09 569	A	CRANKSHAFT REPLACEMENT REPLACE CRANKSHAFT	N/A	X	2
64-16-05 298	Inactive	AC FUEL PUMP OIL SEAL NEW CONF. FUEL PUMP INSTALLED	N/A	X	2
66-06-03 302	A	REPLACEMENT OF CONNECTING ROD ASSY. NEW ROD CONFIGURATION	N/A	X	2
66-20-04 307		OIL FILTER ADAPTER GASKET NEW CONF. GASKET INSTALLED	N/A	X	2
67-22-06 305	4 B	REPLACEMENT OF BENDIX FUEL DIAPHRAGM ASSY NEW OR NEWLY O/H INSTALLED	N/A	X	2
78-23-10 428		BENDIX FUEL INJ. BULLETIN RS-42 NEW P/N INSTALLED	N/A	X	2
79-04-05 433	A	BENDIX FUEL INJ. BULLETIN RS-57 NEW P/N INSTALLED	N/A	X	2
87-10-06 477	R1 A	INSPECTION AND REWORK OF LW-18790 ROCKER ARM ASSY NEW CONFIGURATION P/N INST.	N/A	X	2
90-04-06 488	R1 A	PROPELLER GOVERNOR LINE SUPPORT NEW LINE W/STEEL CON. NUTS INS	N/A	X	2
92-12-05 501	B	LW-14077 PISTON PIN NEW PARTS INSTALLED	N/A	X	2
93-14-15 N/A		INST OF PLCKRD DENOTING OPER. NOT TO EXCEED 33.#HG FIELD COMPLIANCE AFTER STC	N/A	X	2
95-07-01 N/A		CONNECTING ROD BOLT FAILURE LYCOMING PART INSTALLED	N/A	X	2
95-26-02 398		ENGINES OPERATED WITH LOW OCTANE FUEL ENGINE IS NEW OR HAS BEEN REBUILT AT MANUFACTURER	N/A	X	2
98-17-11 N/A		CRANKSHAFT REPAIRED BY NELSON BALANCING SERVICE MAG/PARTICLE INSP OF C/SHAFT	N/A	X	2

Codes:

- 1 AD Applicable to Engine
- 2 AD Not Applicable to Engine
- 3 Field Compliance where applicable

